

## IN THE UNITED STATES PATENT AND TRADEMARK OFFICE

Applicant:

David C. Morris

Group Art Unit:

3745

Serial No.:

09/328,931

Examiner:

C. Verdier

Filed:

06/09/1999

Attorney Docket: MO1.003

Title:

HELICOPTER BLADE ASSEMBLY ADAPTED TO PERMIT

RAPID FORWARD FLIGHT

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## DECLARATION UNDER \$1.132

I have examined the subject concept and provide the following comments:

10 During rapid forward flight the blade assembly of the invention operates in a different manner from that of a standard helicopter.

The standard helicopter works in such a manner that each rotating blade is at an angle of attack relative to the disc swept by its blades and provides lift.

In the invention however the blades are not at an angle of attack relative to the disk swept by its blades. However the disc is at an angle of attack with respect to the forward velocity. At sufficiently high rotation rates, air flows over a virtual disk created by the rotation of the blade assembly to create lift. Lift is generated due to the angle of attack of the total disc, not the angle of attack of each blade as with a helicopter.

Lift will be generated as a function of speed of rotation, angle of attack of the virtual disk and forward speed.

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